

#### **University of Stuttgart**

Institute of Nuclear Technology and Energy Systems

Simulation and Analysis of a Self-Propelling Heat Removal System using sCO<sub>2</sub> at Different Ambient Temperatures M. Hofer, H. Ren, F. Hecker, M. Buck, D. Brillert, J. Starflinger

**KE** 

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## Outline

- 1. Introduction
- 2. Design, layout, control and modelling of sCO<sub>2</sub>-cycle
- 3. Simulation and analysis
- 4. Conclusion and future work

# Introduction (1): Overview and motivation

- Motivation
  - Fukushima
  - Scientific Trend: new heat removal systems
- Active Heat Removal System with Turbo-Compressor
  - sCO<sub>2</sub> as a working fluid
  - Air as ultimate heat sink
  - Self-propelling
  - Very compact

# Introduction (2): Concept of heat removal system



# Introduction (3): Objective of this presentation

- sCO<sub>2</sub>-Brayton-cycle:
  - Thermodynamic Design
  - Layout and control
  - Turbomachinery performance maps
  - Modelling and control of UHS
- Simulation
  - Start-up
  - Varying boundary conditions (especially ambient temperatures)

Modelling and simulations are performed with the thermal-hydraulic system code ATHLET (Analysis of THermal-hydraulics of LEaks and Transients)



# Thermodynamic Design

- Assumptions
  - $\dot{Q}_{CHX} = 10 \text{ MW}$
  - $T_{air} = 45 \ ^{\circ}\mathrm{C}$
  - *T<sub>steam,sat</sub>* = 296.5 °C
  - $\eta_{is,c,t} = 70$  %
  - $\pi_c = 1.7$
- Optimization of compressor inlet pressure p<sub>1</sub> for highest excess power ΔP = P<sub>t</sub> - P<sub>c</sub> - P<sub>fan</sub>
- Result
  - $\Delta P = 283 \text{ kW} (\eta = 2.8 \%)$
  - *p*<sub>1</sub> = 12.6 MPa
  - Considerably above critical point of CO<sub>2</sub>

#### Cycle excess power as a function of compressor inlet pressure



### **Detailed Layout and Control**

- Bypasses
  - Turbine bypass
  - Compressor recirculation
  - UHS bypass
- Control
  - Compressor inlet temperature T<sub>1</sub>
    via fan speed
  - Turbine inlet temperature T<sub>3</sub>
    via turbomachinery shaft speed



#### **Turbomachinery Performance Maps**

Compressor type 1:

 $DP_{comp} = DP_{cycle}$  $DP_{comp} \neq DP_{cvcle}$ 25251.0-©rotational speed in krpm Totational speed in krpm 30 ---surge line ---surge line 20**O** design point of compressor 20**O** design point of compressor  $\mathbf{x}$  design point of cycle  $\mathbf{x}$  design point of cycle  $\Delta h_{t,is} ext{ in } kJ/kg$ 01 21  $\begin{array}{c} \Delta h_{t,is} \text{ in } kJ/kg \\ 0 & \text{cf} \\ 0 & \text{cf} \end{array}$ 25 ઝ ન્ટુ 200 × 铅 3 30 20 20 153 55P 15 10 10 5 0 0 0 2040 60 80 20 4060 80 0  $\dot{m}$  in kg/s $\dot{m}$  in kg/s

Compressor type 2:

# UHS (1): Modelling

- Representative pipe with heat transfer, pressure drop via correlations
- Air side heat transfer correlation validated and extended for low Reynolds numbers with experimental data



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# UHS (2): Modelling

- · Fans not modelled explicitly
- Fan power derived from proportional relationship:  $P_{fan} \sim \frac{\dot{m}^3}{\tilde{\rho}^2}$



# UHS (3): Control

- Motivation
  - Enable cycle operation for the whole range of ambient temperatures from –45 °C to 45 °C
  - Avoid subcritical (two-phase) conditions in the cycle
  - Avoid compressor surge
- Control of compressor inlet temperature T<sub>1</sub>
  - PI-controller:  $\dot{m}_{air}(t) = \dot{m}_{air}(t_0) + K_p^* \Delta T_1(t) + K_i^* \int_{t_0}^t \Delta T_1(\tau) d\tau$
  - Step increase of CO<sub>2</sub> mass flow rate
  - First proportional gain  $K_p^*$  is determined
  - Second integral gain  $K_i^*$  is determined with selected  $K_p^*$

# UHS (4): Control

- Tuning at design ambient temperature of 45 °C (figure)
- Tuning at -45 °C: gains should be 10 times lower



### Simulation (1): Cycle performance

Excess power of cycle at  $T_{comp,in} = 55$  °C and  $T_{air,in} = 45$  °C



#### Simulation (2): Cycle performance

Excess power of cycle at  $T_{comp,in} = 35$  °C and  $T_{air,in} = 25$  °C



#### Simulation (3): Cycle performance

Excess power of cycle at  $T_{comp,in} = 55$  °C and  $T_{air,in} = -45$  °C



#### Start-up of cycle

- Cycle must be operable at any ambient temperature: -45 °C to 45 °C
- (Fast) start-up from cold shutdown conditions might not be possible
  - Material stress
  - Compressor surge
  - Fluid accumulates in UHS
- Alternative: Operational readiness state (ORS)
  - Cycle in part-load during normal operation of nuclear power plant
  - Self-propelling ORS might be possible at only 12 % of design thermal power

#### **NPP simulation: First results**

Excess power of cycle at  $T_{comp,in} = 55$  °C and  $T_{air,in} = 45$  °C



#### Conclusion

- Design, layout, control and modelling of sCO<sub>2</sub> heat removal system
- Start-up from operational readiness state
- Type 2 turbomachinery preferred due to higher surge margin
- Compressor inlet temperature should always be kept constant at 55 °C
- Future Work
  - Further improvement of component models
  - Simulation of the sCO<sub>2</sub>-HeRo-System attached to the NPP (in progress)

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# Thank you!



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# Thermodynamic Design



- Design parameters chosen to guarantee nominal heat removal (10 MWth) at conservatively high ambient temperature (45°C)
- Major criterion: heat removal; efficiency of minor importance
- Takes into account technical design constraints identified in cooperation with WP4 (e.g. maximum temperature difference in HXs)

### **Turbomachinery Performance Maps**

Compressor type 1:  $DP_{comp} = DP_{cycle}$ 



### **Turbomachinery Performance Maps**

Compressor type 2:  $DP_{comp} \neq DP_{cycle}$ 



# UHS (1): Modelling

- Representative pipe with heat transfer, pressure drop via correlations
- Experimental data was used to fit air side heat transfer for low Reynolds numbers
- Fans not modelled explicitly
- Fan power derived from proportional relationship:  $P_{fan} \sim \frac{\dot{m}^3}{\tilde{\sigma}^2}$
- Fans for NPP-UHS are located at the bottom:  $\dot{V} = \frac{\dot{m}}{\rho_{in}}$
- More detailed:  $P_{fan} \sim \Delta p \dot{V} \sim \frac{\dot{m}^2}{0.5(\rho_{in} + \rho_{out})} \frac{\dot{m}}{\rho_{in}}$
- Design point power is calculated assuming a specific power consumption

$$\Delta P_{fan,des} = \dot{Q}_{UHS} * 8.5 \text{ kW}_{el}/\text{MW}_{th}$$

# UHS (2): Control

Compressor inlet temperature vs. density control

- Operation points considerably above critical point → gradients in thermodynamic properties are lower
- No inventory control
  - Pressures and temperatures are linked
  - E.g. when air flow rate is increased, compressor inlet temperature and pressure will decrease
  - Resulting in almost constant compressor inlet density
  - Difficult to use density as a control parameter

## Start-up of cycle

	Unit	ORS1	ORS2	ORS3
Turbomachinery speed relative to the cycle design point	%	20	20	20
Turbine bypass valve	%	58	24	0
Turbine valves	%	0	0	100
Compressor inlet p	bar	117.3	122.1	122.6
Compressor outlet p	bar	119.4	125.2	125.7
CHX outlet T	°C	111	150	155
CHX thermal power	MW	1.2	1.2	1.2
Mass flow rate (CO <sub>2</sub> )	kg/s	8.5	5.9	5.7
Compressor efficiency	%	50.7	68.2	68.9
Turbine efficiency	%	0	0	71.4
Compressor power	kW	7.1	5.0	4.9
Turbine power	kW	0	0	6.5
Fan power	kW	0.6	0.4	0.4
Total power	kW	-7.7	-5.4	1.2

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## Simulation (4): UHS-QT-diagram

• 
$$T_{comp,in} = 55 \text{ °C}, T_{air,in} = 45 \text{ °C}$$
 vs.  $T_{comp,in} = 35 \text{ °C}, T_{air,in} = 25 \text{ °C}$ 



#### Simulation (5): UHS-subvolumes-T-diagram

• 
$$T_{comp,in} = 55 \text{ °C}, T_{air,in} = 45 \text{ °C}$$
 vs.  $T_{comp,in} = 35 \text{ °C}, T_{air,in} = 25 \text{ °C}$ 



#### NPP simulation: First results • $T_{comp,in} = 55$ °C and $T_{air,in} = 45$ °C Cool-down to $T_3 = 260 \text{ °C}$ $\times 10^4$ 0.25 2.5 $\Delta P < 0$ 0.2 2 nim/1 ni n 1'2 ∆P in MW DP 0.15 MD 0.1 $T_3 > T_{max}$ 1 0.05 Control of *n* to keep $T_3$ constant From ORS 0.5 8 10 4 6 $\boldsymbol{Q}_{\text{CHX}}$ in MW

#### **NPP simulation: First results**

4 systems (with adaption to decay heat curve)

- Control of turbomachinery speed and subsequent shutdown
- Systems can run for more than 72 h
- Excess power always higher than zero

